Date: 21.12.2011

Subject: Investigation Report on EC135 P2+ S/N 0583

Aircarft:

EC135 P2+ S/N: 583

Reg: YR-CPC

F/H: 1.378:45 LDG: 2601

(Flight&Technical Log No.E 465 28.11.2011) (Flight&Technical Log No.E 465 28.11.2011)

Operator:

C&I Corporation

Industillor Street no.1 601124 Onesti-Bacau

Romania

**Equipment:** 

CPDS; FCDS; AFCS; EURONAV IV

ATC Flight number:

0212032127 28.11.2011

OITUZ (DEP) to GHERGHITA (ARZ)

Crew:

PIC (R/H) Valentin Stoica,

2 PAX seated rear L/H and R/H, looking in flight direction

Log Discrepancy: Flight & Technical Log No. E 0465

malfunction of autopilot,

caution displayed "YAW SAS" and "ACTUATION", following by a brutal movement to the right

following by a brutal movement to the right with aggressive roll and descenting with diving

Exceedance: - mast moment exceeding

MM >66% 10,4 sec MM >78% 5,4 sec

- engine overtorque

- ENG ITRQ 127 % ca. 10 sec higher than 82%

- ENG II TRQ 127 % ca. 10 sec higher than 82%

rotor over speed
NR >130%

and the same that we will be the first

## Actions performed on aircraft for investigation (19.12. - 20.12.2011):

check of aircraft log books

- record of CPDS FLIGHT REPORT (pictures)

- record of CPDS MAINTENANCE REPORT (pictures)

attempt to download flight path from EuroNav IV

 Autopilot Installation Acceptance Test Procedure performed according to L221V2101 X04 Issue C

- AFCS actuator tests (EHAs, SEMAs, TRIMs)

visual inspection without removal of parts



## Findings (19.12. – 20.12.2011):

- damage through contact of all blades control cuffs to Mast
- damage through contact of all blades control cuffs to Flex Beam
- pilot (R/H) front window has scratches inside
- pilot (R/H) front lower front window shoe marks inside
- wiper in L/H end position (was not in use)
- one weather radar mounting rivet head is broken
- A.TRIM button surface impressed on Autopilot Mode Selector (function not affected)
- autopilot tests without findings
- EURONAV IV system did not work at all, flight path download not feasible

## Pilot interview:

The pilot confirmed the basic information already received via ECROM and following aspects were clarified:

- pilot was flying hands-on the cyclic control due to possibility of further turbulence
- co-pilot seat vacant
- heading 187°
- initial movement: brutal yawing to the right
- followed by roll right over 90° bank and nose down
- entering into spinning motion ("like tail rotor drive shaft broken")
- autopilot upper modes decoupled via cyclic grip switch APMD DCPL
- flight to Baneasa airport performed with autopilot engaged (upper modes off)

## Actions required from Eurocopter Deutschland GmbH:

- check of engine restart (ground idle only) to preserve main gear box and tail gear box according AMM (aircraft maintenance manual)
- engine I + II have to be preserved, removed and send for investigation/repair to manufacture P&W
- main rotor blades have to be removed and to be send for investigation/repair to the manufacture Eurocopter Deutschland
- EuroNav IV hard drive has to be send for "read out" to BFU Braunschweig / Germany
- inspections on airframe and dynamic components have to be performed according ETA (Eurocopter Technical Agreement) -05-105

Peter Kalmutchi

Aeronautical Inspector Engineering Investigator CIAS Romania

**Robert Schmelz** 

Eurocopter Technical Expert Electrics/Avionic

**Christian Melz** 

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Sören Ilgen

Eurocopter Product Manager EC135/635&BO105

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